

**NOTICE**

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

**8. Description of Work Accomplished**

(if more space is required, attach additional sheets, identify with aircraft nationality and registration mark and date work completed.)

REMOVED FACTORY INSTALLED SEAT, SEATBACK AND SEATBACK ADJUSTING BRACKETS WITHOUT DISTURBING THE SEAT BELT MOUNTING ON THE AIRFRAME.

INSTALLED LEFT HAND SEAT ASSEMBLY PART NUMBER 0400118-65 AND RIGHT HAND SEAT ASSEMBLY PART NUMBER 0400118-66. INSTALLED TWO INBOARD SEAT RAILS PART NUMBER 0411545-1 AND TWO OUTBOARD SEAT RAILS PART NUMBER 0411545-2.

THE SEAT RAILS WERE INSTALLED ON TOP OF 3/8 ALUMINUM STOCK 1 1/2 INCHES BY 1 1/2 INCHES WITH MS27039 STRUCTURAL SCREWS AND MS21044N3 NUTS. A 1 1/2 INCH BY 2 1/4 INCH DOUBLER OF .060 ALUMINUM WAS FABRICATED AND FASTENED BENEATH THE FLOOR PAN OF THE AIRCRAFT BENEATH EACH SEAT RAIL FOR ADDED STRENGTH. EACH SEAT RAIL WAS SECURED TO THE FLOOR PAN ASSEMBLY WITH EIGHT MS27039-1-15 STRUCTURAL SCREWS AND MS21044N3 NUTS. SEAT STOPS, CESSNA PART NUMBER 0713074-1, WERE INSTALLED AT FRONT AND REAR OF EACH OUTBOARD RAIL.

THE SEAT BELT BRACKET INSTALLATION TO THE AIRFRAME WAS NOT ALTERED FROM THE ORIGINAL FACTORY INSTALLATION.

PREVIOUS 337 INSTALLATION DATA DATED 12-4-86, SIGNED M.O. PETERSON, AGL-FSD061, REFERED TO AND USED AS A GUIDE FOR THIS INSTALLATION.

WEIGHT AND BALANCE REVISED

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Additional Sheets Are Attached